CESSNA MODEL 182T NAV III GFC 700 AFCS SECTION 4 NORMAL PROCEDURES

# **NORMAL PROCEDURES**

# **TABLE OF CONTENTS**

P	age'
Introduction	.4-3
Airspeeds For Normal Operation	.4-3
NORMAL PROCEDURES	
Cabin	.4-5
Empennage	.4-6
Right Wing	
Left Wing Leading Edge	.4-9
Left Wing	4-10
Before Starting Engine	
Starting Engine (With External Power)	4-13
Takeoff	4-18
Normal Takeoff	
Enroute Climb	
Maximum Performance Climb	4-19
Descent	4-20
Before Landing	
Normal Landing	
Balked Landing	4-22
After Landing	4-22 4-22

### CESSNA MODEL 182T NAV III GFC 700 AFCS

# TABLE OF CONTENTS (Continued)

	Page
AMPLIFIED NORMAL PROCEDURES	4-23
Preflight Inspection	4-23
Starting Engine	
Recommended Starter Duty Cycle	
Leaning For Ground Operations	
Taxiing	
Before Takeoff	
Warm Up	
Magneto Check	
Alternator Check	
Elevator Trim	
Landing Lights	
Takeoff	
Power Check	
Wing Flap Settings.	
Crosswind Takeoff	
Enroute Climb.	
Cruise	
Leaning Using Exhaust Gas Temperature (EGT)	
Fuel Varian Procedures For Normal Operations	
Fuel Vapor Procedures	
Stalls	
Landing	
Normal Landing	
Short Field Landing	
Crosswind Landing	
Balked Landing	
Cold Weather Operations	
Starting	
Winterization Kit	
Hot Weather Operations	
Noise Characteristics	. 4-46

# INTRODUCTION

Section 4 provides procedures and amplified instructions for normal operations using standard equipment. Normal procedures associated with optional systems can be found in Section 9, Supplements.

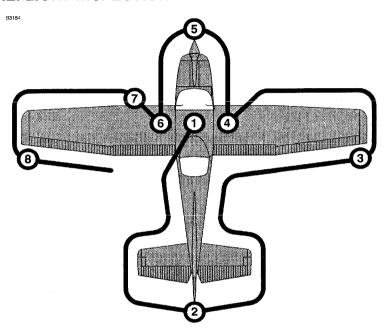
### AIRSPEEDS FOR NORMAL OPERATION

Unless otherwise noted, the following speeds are based on a maximum weight of 3100 pounds and may be used for any lesser weight.

TAKEOFF	_
Normal Climb	S
ENROUTE CLIMB, FLAPS UP  Normal, Sea Level	0
Best Rate of Climb, Sea Level	
Best Rate of Climb, 10,000 Feet	
Best Angle of Climb, Sea Level	
Best Angle of Climb, 10,000 Feet	S
LANDING APPROACH  Normal Approach, Flaps UP	S
BALKED LANDING	
Maximum Power, Flaps 20°55 KIA	S
MAXIMUM RECOMMENDED TURBULENT AIR PENETRATION SPEED	₹
3100 POUNDS	
2600 POUNDS	
2100 POUNDS	S
MAXIMUM DEMONSTRATED CROSSWIND VELOCITY Takeoff or Landing	S
	_

## **NORMAL PROCEDURES**

### PREFLIGHT INSPECTION



#### NOTE

Visually check airplane for general condition during walk-around inspection. Airplane should be parked in a normal ground attitude (refer to Figure 1-1) to make sure that fuel drain valves allow for accurate sampling. Use of the refueling steps and assist handles will simplify access to the upper wing surfaces for visual checks and refueling operations. In cold weather, remove even small accumulations of frost, ice or snow from wing, tail and control surfaces. Also, make sure that control surfaces contain no internal accumulations of ice or debris. Prior to flight, check that pitot heater is warm to touch within 30 seconds with battery and pitot heat switches on. If a night flight is planned, check operation of all lights, and make sure a flashlight is available.

Figure 4-1

CESSNA MODEL 182T NAV III GFC 700 AFCS SECTION 4 NORMAL PROCEDURES

# Fluids:

ORM

Sump Drains - 13 Dip Tanks Oil Dip

# PREFLIGHT INSPECTION (Continued)

#### ① CABIN

- 1. Pitot Tube Cover REMOVE (check for pitot blockage)
- 2. Pilot's Operating Handbook ACCESSIBLE TO PILOT
- Garmin G1000 Cockpit Reference Guide ACCESSIBLE TO
  PILOT
- 4. Airplane Weight and Balance CHECKED
- 5. Parking Brake SET
- 6. Control Wheel Lock REMOVE

"Balanced Field Length" take off to 50' + land from 50'

#### WARNING

WHEN THE MASTER SWITCH IS ON, USING AN EXTERNAL POWER SOURCE, OR MANUALLY ROTATING THE PROPELLER, TREAT THE PROPELLER AS IF THE MAGNETOS SWITCH WERE ON. DO NOT STAND, NOR ALLOW ANYONE ELSE TO STAND, WITHIN THE ARC OF THE PROPELLER SINCE A LOOSE OR BROKEN WIRE, OR A COMPONENT MALFUNCTION, COULD CAUSE THE ENGINE TO START.

- 7. MAGNETOS Switch OFF
- 8. AVIONICS Switch (BUS 1 and BUS 2) OFF
- 9. MASTER Switch (ALT and BAT) ON
- 10. Primary Flight Display (PFD) CHECK (verify PFD is ON)
- 11.FUEL QTY (L and R) CHECK
- 12.LOW FUEL L and LOW FUEL R Annunciators CHECK (verify annunciators are not shown on PFD)
- 13.OIL PRESSURE Annunciator CHECK (verify annunciator is shown)
- 14.LOW VACUUM Annunciator CHECK (verify annunciator is shown)
- 15.AVIONICS Switch (BUS 1) ON
- 16. Forward Avionics Fan CHECK (verify fan is heard)

Hobbs/Tach Engine/System/Gal Rem Lights (Nav/Rot/Strobe/Pulse)

### ① CABIN (Continued)

- 17.AVIONICS Switch (BUS 1) OFF
- 18.AVIONICS Switch (BUS 2) ON
- 19.Aft Avionics Fan CHECK (verify fan is heard)
- 20.AVIONICS Switch (BUS 2) OFF
- 21.PITOT HEAT Switch ON (carefully check that pitot tube is warm to the touch within 30 seconds)
- 22.PITOT HEAT Switch OFF
- 23.Stall Warning System CHECK (gently move the stall vane upward and verify that the stall warning horn is heard)
- 24.LOW VOLTS Annunciator CHECK (verify annunciator is shown)
- 25.MASTER Switch (ALT and BAT) OFF
- 26. Elevator and Rudder Trim Controls TAKEOFF position
- 27.FUEL SELECTOR Valve BOTH
- 28.ALT STATIC AIR Valve OFF (push full in)
- 29. Fire Extinguisher CHECK (verify gage pointer in green arc)

### 2 EMPENNAGE

- 1. Baggage Compartment Door CHECK (lock with key)
- 2. Rudder Gust Lock (if installed) REMOVE
- 3. Tail Tiedown DISCONNECT
- 4. Control Surfaces CHECK (freedom of movement and security)
- 5. Trim Tabs CHECK (security)
- 6. Antennas CHECK (security of attachment and general condition)

# ③ RIGHT WING Trailing Edge

- 1. Flap CHECK (security and condition)
- 2. Aileron CHECK (freedom of movement and security)

#### RIGHT WING

- 1. Wing Tiedown DISCONNECT
- 2. Fuel Tank Vent Opening CHECK (verify opening is clear)
- 3. Main Wheel Tire CHECK (proper inflation and general condition (weather checks, tread depth and wear, etc.))
- 4. Fuel Tank Sump Quick Drain Valves DRAIN

  Drain at least a cupful of fuel (using sampler cup) from each sump location to check for water, sediment, and proper fuel grade before each flight and after each refueling. If water is observed, take further samples until clear and then gently rock wings and lower tail to the ground to move any additional contaminants to the sampling points. Take repeated samples from all fuel drain points until all contamination has been removed. If contaminants are still present, refer to WARNING below and do not fly airplane.

#### NOTE

Collect all sampled fuel in a safe container. Dispose of the sampled fuel so that it does not cause a nuisance, hazard or damage to the environment.

#### WARNING

IF, AFTER REPEATED SAMPLING, EVIDENCE OF CONTAMINATION STILL EXISTS, THE AIRPLANE SHOULD NOT BE FLOWN. TANKS SHOULD BE DRAINED AND SYSTEM PURGED BY QUALIFIED MAINTENANCE PERSONNEL. ALL EVIDENCE OF CONTAMINATION MUST BE REMOVED BEFORE FURTHER FLIGHT.

- 5. Fuel Quantity CHECK VISUALLY (for desired level)
- Fuel Filler Cap SECURE and VENT CLEAR

### **S** NOSE

- Static Source Opening (right side of fuselage) CHECK (verify opening is clear)
- 2. Fuel Strainer Quick Drain Valve (located on lower right side of engine cowling) DRAIN

Drain at least a cupful of fuel (using sampler cup) from valve to check for water, sediment, and proper fuel grade before each flight and after each refueling. If water is observed, take further samples until clear and then gently rock wings and lower tail to the ground to move any additional contaminants to the sampling points. Take repeated samples from all fuel drain points, including the fuel return line and fuel selector, until all contamination has been removed. If contaminants are still present, refer to WARNING below and do not fly the airplane.

#### **NOTE**

Collect all sampled fuel in a safe container. Dispose of the sampled fuel so that it does not cause a nuisance, hazard, or damage to the environment.

#### WARNING

IF, AFTER REPEATED SAMPLING, EVIDENCE OF CONTAMINATION STILL EXISTS, THE AIRPLANE SHOULD NOT BE FLOWN. TANKS SHOULD BE DRAINED AND SYSTEM PURGED BY QUALIFIED MAINTENANCE PERSONNEL. ALL EVIDENCE OF CONTAMINATION MUST BE REMOVED BEFORE FURTHER FLIGHT.

3. Engine Cooling Air Inlets - CHECK (clear of obstructions)

- ⑤ NOSE (Continued)
- Propeller and Spinner CHECK (for nicks, security and no red oil leaks)

#### **NOTE**

Minor leaking of the blade seal area is possible on new propellers as the seals wear in. Any initial leakage will be visible as minor streaking on the blade or blades. Clean off oil residue and cycle propeller at least 5 times. Oil leakage should be reduced or completely stopped. If minor leaking continues after 20 hours of operation or increases remove propeller and have repaired.

- 5. Air Filter CHECK (for restrictions by dust or other foreign matter)
- Nosewheel Strut and Tire CHECK (proper inflation of strut and general condition of tire (weather checks, tread depth and wear, etc.))
- 7. Engine Oil Dipstick/Filler Cap:
  - a. Oil level CHECK
  - b. Dipstick/filler cap SECURE

#### NOTE

**Do not operate with less than 4 quarts.** Fill to 9 quarts for extended flight.

8. Static Source Opening (left side of fuselage) - CHECK (verify opening is clear)

# **6** LEFT WING Leading Edge

- 1. Fuel Tank Vent Opening CHECK (blockage)
- 2. Stall Warning Vane CHECK (freedom of movement)
- Landing/Taxi Light(s) CHECK (condition and cleanliness of cover)

#### ⑦ LEFT WING

- 1. Wing Tiedown DISCONNECT
- 2. Fuel Quantity CHECK VISUALLY (for desired level)
- 3. Fuel Filler Cap SECURE and VENT CLEAR
- 4. Fuel Tank Sump Quick Drain Valves DRAIN

Drain at least a cupful of fuel (using sampler cup) from each sump location to check for water, sediment, and proper fuel grade before each flight and after each refueling. If water is observed, take further samples until clear and then gently rock wings and lower tail to the ground to move any additional contaminants to the sampling points. Take repeated samples from all fuel drain points until all contamination has been removed. If contaminants are still present, refer to WARNING below and do not fly airplane.

#### NOTE

Collect all sampled fuel in a safe container. Dispose of the sampled fuel so that it does not cause a nuisance, hazard, or damage to the environment.

#### WARNING

IF, AFTER REPEATED SAMPLING, EVIDENCE OF CONTAMINATION STILL EXISTS, THE AIRPLANE SHOULD NOT BE FLOWN. TANKS SHOULD BE DRAINED AND SYSTEM PURGED BY QUALIFIED MAINTENANCE PERSONNEL. ALL EVIDENCE OF CONTAMINATION MUST BE REMOVED BEFORE FURTHER FLIGHT.

5. Main Wheel Tire - CHECK (proper inflation and general condition (weather checks, tread depth and wear, etc.))

# LEFT WING Trailing Edge

- 1. Aileron CHECK (freedom of movement and security)
- 2. Flap CHECK (security and condition)

### SECTION 4 NORMAL PROCEDURES

cranked to pilot's height

**Seat Adjusted/** 

### **BEFORE STARTING ENGINE**

- 1. Preflight Inspection COMPLETE
- 2. Passenger Briefing COMPLETE
- 3. Seats and Seat Belts ADJUST and LOCK (verify inertia reel locking)
- 4. Brakes TEST and SET
- 5. Circuit Breakers CHECK IN
- 6. Electrical Equipment OFF
- 7. AVIONICS Switch (BUS 1 and BUS 2) OFF

#### **CAUTION**

THE AVIONICS SWITCH (BUS 1 AND BUS 2) MUST BE OFF DURING ENGINE START TO PREVENT POSSIBLE DAMAGE TO AVIONICS.

- 8. Cowl Flaps OPEN
- 9. FUEL SELECTOR Valve BOTH

# **Passenger Brief**

seat belts & shoulder harness personal electronic devices air vents/heat/comfort fire extinguisher location/operation emergency procedures/exits

#### **Mission Brief**

mission objective
destination, wx, route, alternate, ETE
NOTAMS
crew coordination & CRM
STERILE COCKPIT PROCEDURES
cockpit layout
intercom & radio usage
seats/ seatbelts/ doors
emergency action & equipment

# STARTING ENGINE (With Battery)

- Throttle Control OPEN 1/4 INCH
- 2. Propeller Control HIGH RPM (push full in)
- Mixture Control IDLE CUTOFF (pull full out)
- 4. STBY BATT Switch:
  - TEST (hold for 10 seconds, verify that green TEST lamp does not go off)
  - b. ARM (verify that PFD comes on)
- Engine Indicating System CHECK PARAMETERS (verify no red X's through ENGINE page indicators)
- 6. BUS E Volts CHECK (verify 24 VOLTS minimum shown)
- M BUS Volts CHECK (verify 1.5 VOLTS or less shown)
- 8. BATT S Amps CHECK (verify discharge shown (negative))
- 9. STBY BATT Annunciator CHECK (verify annunciator is shown)
- Propeller Area CLEAR (verify that all people and equipment are at a safe distance from the propeller)
- 11. MASTER Switch (ALT and BAT) ON
- 12. BEACON Light Switch ON

# 12A Headsets On NOTE

If engine is warm, omit priming procedure steps 13 thru 15 below.

- 13. FUEL PUMP Switch ON
- 14. Mixture Control SET to FULL RICH (full forward) until stable fuel flow is indicated (approximately 3 to 5 seconds), then set to IDLE CUTOFF (full aft) position.
- 15. FUEL PUMP Switch OFF
- 16. MAGNETOS Switch START (release when engine starts)
- Mixture Control ADVANCE SMOOTHLY TO RICH (when engine starts)

#### NOTE

If the engine is primed too much (flooded), place the mixture control in the IDLE CUTOFF position, open the throttle control 1/2 to full, and engage the starter motor (START). When the engine starts, advance the mixture control to the FULL RICH position and retard the throttle control promptly.

# STARTING ENGINE (With Battery) (Continued)

- Oil Pressure CHECK (verify that oil pressure increases into the GREEN BAND range in 30 to 60 seconds)
- AMPS (M BATT and BATT S) CHECK (verify charge shown (positive))
- LOW VÓLTS Annunciator CHECK (verify annunciator is not shown)
- 21. NAV Light Switch ON as required
- 22. AVIONICS Switch (BUS 1 and BUS 2) ON
- 23. Lean 1200 RPM to highest RPM
- 24. Mission Master ON
- 25. GPS verity database & accept
- 26. Frequencies SET
- 27. ATIS/ASOS copy
- 28. Engine/System/Fuel verify total
- 29. Field Diagram or Safe Taxi in view

New Century Ground CAPFLIGHT 1497 Taxi from the North T hangars with the 1 minute Wx VFR to the SW

### BEFORE TAKEOFF

- 1. Parking Brake SET
- 2. Pilot and Passenger Seat Backs MOST UPRIGHT POSITION
- 3. Seats and Seat Belts CHECK SECURE
- 4. Cabin Doors CLOSED and LOCKED
- 5. Flight Controls FREE and CORRECT
- 6. Flight Instruments (PFD) CHECK (no red X's)
- 7. Altimeters:
- a. PFD (BARO) SETb. Standby Altimeter SET
- 8. ALT SEL SET
- 9. Standby Flight Instruments CHECK10. Fuel Quantity CHECK (verify level is correct)

#### NOTE

Flight is not recommended when both fuel quantity indicators are in the yellow band range.

- 11. Mixture Control RICH
- 12. FUEL SELECTOR Valve SET BOTH
- 13. Autopilot ENGAGE (push AP button on either PFD or MFD
  - bezel) L Flight Controls - CHFCK
  - 14. Flight Controls CHECK (verify autopilot can be overpowered in both pitch and roll axes)

(Continued Next Page)

182TPHBUS-01

U.S.

4-15

### **BEFORE TAKEOFF** (Continued)

- 15. A/P TRIM DISC Button PRESS (verify autopilot disengages and aural alert is heard)
- 16. Flight Director OFF (push FD button on either PFD or MFD bezel)
  - 17. Elevator and Rudder Trim Controls SET FOR TAKEOFF
  - 18. Throttle Control 1800 RPM
    - MAGNETOS Switch CHECK (RPM drop should not exceed 175 RPM on either magneto or 50 RPM differential between magnetos)
    - b. Propeller Control CYCLE (from high to low RPM; return to high RPM) (push full in)
    - c. VAC Indicator CHECK
    - d. Engine Indicators CHECK
    - e. Ammeters and Voltmeters CHECK
  - 19. Annunciators CHECK (verify no annunciators are shown)
  - 20. Throttle Control CHECK IDLE
  - 21. Throttle Control 1000 RPM or LESS
  - 22. Throttle Control Friction Lock ADJUST
  - 23. COM Frequency(s) SET
  - 24. NAV Frequency(s) SET
  - 25. FMS/GPS Flight Plan AS DESIRED

#### NOTE

Check GPS availability on AUX-GPS STATUS page. No annunciation is provided for loss of GPS2. (AUX page 3)

26. XPDR - SFT

### **BEFORE TAKEOFF** (Continued)

27. CDI Softkey - SELECT NAV SOURCE

#### CAUTION

THE G1000 HSI SHOWS A COURSE DEVIATION INDICATOR FOR THE SELECTED GPS, NAV 1 OR NAV 2 NAVIGATION SOURCE. THE G1000 HSI DOES NOT PROVIDE A WARNING FLAG WHEN A VALID NAVIGATION SIGNAL IS NOT BEING SUPPLIED TO THE INDICATOR. WHEN A VALID NAVIGATION SIGNAL IS NOT BEING SUPPLIED, THE COURSE DEVIATION BAR (D-BAR) PART OF THE INDICATOR IS NOT SHOWN ON THE HSI COMPASS CARD. THE MISSING D-BAR IS CONSIDERED TO BE THE WARNING FLAG.

#### WARNING

WHEN THE AUTOPILOT IS ENGAGED IN NAV, APR OR BC OPERATING MODES, IF THE HSI NAVIGATION SOURCE IS CHANGED MANUALLY, USING THE CDI SOFTKEY, THE CHANGE WILL INTERRUPT THE NAVIGATION SIGNAL TO THE AUTOPILOT AND WILL CAUSE THE AUTOPILOT TO REVERT TO ROL MODE OPERATION. NO AURAL ALERT WILL BE PROVIDED. IN ROL MODE. THE AUTOPILOT WILL ONLY KEEP THE WINGS LEVEL AND WILL NOT CORRECT THE AIRPLANE HEADING OR COURSE. SET THE HDG BUG TO THE CORRECT HEADING AND SELECT THE CORRECT NAVIGATION SOURCE ON THE HSI, USING THE CDI SOFTKEY. BEFORE **ENGAGING** AUTOPILOT IN ANY OTHER OPERATING MODE.

- 28. CABIN PWR 12V Switch OFF
- 29. Wing Flaps UP 20° (10° preferred)
- 30. Cowl Flaps OPEN
- 31. Cabin Windows CLOSED and LOCKED
- 32. STROBE Light Switch ON PULSE or Landing Lt ON
- 33. Brakes RELEASE

### 34. Takeoff Brief

#### **TAKEOFF**

#### NORMAL TAKEOFF

- 1. Wing Flaps UP 20° (10° preferred)
- 2. Throttle Control FULL (push full in)
- 3. Propeller Control 2400 RPM
- 4. Mixture Control RICH (above 5000 feet pressure altitude, lean for maximum RPM)
- 5. Elevator Control LIFT NOSEWHEEL (at 50 60 KIAS)
- Climb Airspeed -70 KIAS (FLAPS 20°) 80 KIAS (FLAPS UP)
- 7. Wing Flaps RETRACT (at safe altitude)

#### SHORT FIELD TAKEOFF

- 1. Wing Flaps 20°
- 2. Brakes APPLY
- 3. Throttle Control FULL (push full in)
- 4. Propeller Control 2400 RPM
- Mixture Control RICH (above 5000 feet pressure altitude, lean for maximum RPM)
- 6. Brakes RELEASE
- 7. Elevator Control SLIGHTLY TAIL LOW
- 8. Climb Airspeed 58 KIAS (until all obstacles are cleared)
- Wing Flaps RETRACT SLOWLY (when airspeed is more than 70 KIAS)

# **Takeoff / Climbout Sequence**

Heading Bug - press to center aligned on r/w

Go Around - activate (if desired)

**Power - Engine Instruments/Airspeed Alive** 

Rotate - at tape mark

70 KIAS - Flaps Up

90 KIAS - Power 23" / Fuel Top of Green

Autopilot - stabilize ROC, Trim, Heading Bug centered

> 800'

### **ENROUTE CLIMB**

#### **NORMAL CLIMB**

- 1. Airspeed 85 95 KIAS
- 2. Throttle Control 23 in.hg. or FULL (if less than 23 in.hg.)
- 3. Propeller Control 2400 RPM
- 4. Mixture Control 15 GPH or FULL RICH (if less than 15 GPH)
- 5. FUEL SELECTOR Valve BOTH
- 6. Cowl Flaps OPEN (as required)

#### MAXIMUM PERFORMANCE CLIMB

- Airspeed -80 KIAS at sea level 74 KIAS at 10,000 feet
- 2. Throttle Control FULL (push full in)
- 3. Propeller Control 2400 RPM
- 4. Mixture Control FULL RICH (or SET to Maximum Power Fuel Flow Placard value for altitude in Amplified Normal Procedures)
- 5. FUEL SELECTOR Valve BOTH
- 6. Cowl Flaps OPEN

# **CRUISE**

- Power 15 23 in.hg. at 2000 2400 RPM (no more than 80% power recommended)
- 2. Elevator and Rudder Trim Controls ADJUST
- 3. Mixture Control LEAN (for desired performance or economy)
- 4. Cowl Flaps CLOSED
- 5. FMS/GPS REVIEW and BRIEF (OBS/SUSP softkey operation for holding pattern procedure (IFR))

#### DESCENT

- 1. Power AS DESIRED
- Mixture ADJUST (if necessary to make engine run smoothly)
- 3. Cowl Flaps CLOSED
- 4. Altimeters:
  - a. PFD (BARO) SET
  - b. Standby Altimeter SET
- 5. ALT SEL SET
- 6. CDI Softkey SELECT NAV SOURCE
- 7. FMS/GPS REVIEW and BRIEF (OBS/SUSP softkey operation for holding pattern procedure (IFR))

#### CAUTION

THE G1000 HSI SHOWS A COURSE DEVIATION INDICATOR FOR THE SELECTED GPS, NAV 1 OR NAV 2 NAVIGATION SOURCE. THE G1000 HSI DOES NOT PROVIDE A WARNING FLAG WHEN A VALID NAVIGATION SIGNAL IS NOT BEING SUPPLIED TO THE INDICATOR. WHEN A VALID NAVIGATION SIGNAL IS NOT BEING SUPPLIED, THE COURSE DEVIATION BAR (D-BAR) PART OF THE INDICATOR IS NOT SHOWN ON THE HSI COMPASS CARD. THE MISSING D-BAR IS CONSIDERED TO BE THE WARNING FLAG.

#### WARNING

WHEN THE AUTOPILOT IS ENGAGED IN NAV, APR OR BC OPERATING MODES, IF THE HSI NAVIGATION SOURCE IS CHANGED MANUALLY, USING THE CDI SOFTKEY, THE CHANGE WILL INTERRUPT THE NAVIGATION SIGNAL TO THE AUTOPILOT AND WILL CAUSE THE AUTOPILOT TO REVERT TO ROL MODE OPERATION. NO AURAL ALERT WILL BE PROVIDED. IN ROL MODE, THE AUTOPILOT WILL ONLY KEEP THE WINGS LEVEL AND WILL NOT CORRECT THE AIRPLANE HEADING OR COURSE. SET THE HDG BUG TO THE CORRECT HEADING AND SELECT THE CORRECT NAVIGATION SOURCE ON THE HSI, USING THE CDI SOFTKEY, BEFORE ENGAGING THE AUTOPILOT IN ANY OTHER OPERATING MODE.

- 8. FUEL SELECTOR Valve BOTH
- 9. Wing Flaps AS DESIRED (UP 10° below 140 KIAS)

(10° - 20° below 120 KIAS)

(20° - FULL below 100 KIÁS)

# 10. Landing Weight - Calculate

4-20 U.S. 182TPHBUS-01

### **BEFORE LANDING**

- 1. Pilot and Passenger Seat Backs MOST UPRIGHT POSITION
- 2. Seats and Seat Belts SECURED and LOCKED
- FUEL SELECTOR Valve BOTH
- 4. Mixture Control RICH
- 5. Propeller Control HIGH RPM (push full in)
- LAND and TAXI Light Switches ON
- 7. Autopilot OFF
- 8. CABIN PWR 12V Switch OFF

#### LANDING

#### **NORMAL LANDING**

- 1. Airspeed 70 80 KIAS (Flaps UP)
- 2. Wing Flaps AS DESIRED (UP 10° below 140 KIAS)

(10° - 20° below 120 KIAS)

(20° - FULL below 100 KIAS)

- 3. Airspeed 60 70 KIAS (Flaps FULL)
- 4. Elevator and Rudder Trim Controls ADJUST
- 5. Touchdown MAIN WHEELS FIRST
- 6. Landing Roll LOWER NOSEWHEEL GENTLY
- 7. Braking MINIMUM REQUIRED

#### SHORT FIELD LANDING

- 1. Airspeed 70 80 KIAS (Flaps UP)
- 2. Wing Flaps FULL (below 100 KIAS)
- 3. Airspeed 60 KIAS (until flare)
- 4. Elevator and Rudder Trim Controls ADJUST
- 5. Power REDUCE TO IDLE (as obstacle is cleared)
- 6. Touchdown MAIN WHEELS FIRST
- 7. Brakes APPLY HEAVILY
- 8. Wing Flaps UP

# LANDING (Continued)

#### **BALKED LANDING**

- 1. Throttle Control FULL (push full in) and 2400 RPM
- 2. Wing Flaps RETRACT to 20°
- 3. Climb Speed 55 KIAS
- 4. Wing Flaps RETRACT SLOWLY (after reaching a safe altitude and 70 KIAS)
- 5. Cowl Flaps OPEN

### AFTER LANDING

- 1. Wing Flaps UP
- 2. Cowl Flaps OPEN
- 3. Lean

### **SECURING AIRPLANE**

- 1. Parking Brake SET
- 2. Throttle Control IDLE (pull full out)
- 3. Electrical Equipment OFF
- 4. AVIONICS Switch (BUS 1 and BUS 2) OFF
- 5. Mixture Control IDLE CUTOFF (pull full out)
- 6. MAGNETOS Switch OFF
- 7. MASTER Switch (ALT and BAT) OFF Hobbs/Tach
- 8. STBY BATT Switch OFF
- 9. Control Lock INSTALL
- 10.FUEL SELECTOR Valve LEFT or RIGHT (to prevent crossfeeding between tanks)

### N997CP / CPF1497 Protocol

Scheduling thru a/c manager, LtCol John Shelton, 816.392.1444 (24/7 voice/text). Please advise when back in hangar after usage.

Hangar E14, NEC of IXD airport; hangar access thru Advanced Aviation, a/c key is stored in black a/c notebook on planning table under left wing.

Fuel dip stick behind pilot's seat. Each pilot dips tanks as part of preflight and enters fuel load into Engine>System>. An "annotated check list", adding the items normally found on a CAP checklist, is in the aircraft or at <a href="http://kcscouts.home.att.net/997\_090724.pdf">http://kcscouts.home.att.net/997\_090724.pdf</a>.

On initial call to ground, state location as "North T"

After flight, dip tanks and tell Advanced Aviation (913.768.1500) driver how many gallons to put on each side to bring back to 50 gal ramp load. Get the fuel invoice in AA office for inclusion with 108. Annotate receipt with mission code, mission number, PIC name, hobbs time, and if member paid. (Alternative is to use self service pump at Advanced Aviation. CAP receives self-serve price in any event.) Use of gas card in book is by Wing permission.

A/C is pushed back into hangar using vertical stabilizer sight line on back hangar wall for proper alignment. Check position of main mount stops against orange calibration spots before pushing in since clearance behind horizontal stabilizer is minimal.

Cleaning supplies for a/c are on table behind right wing. Use Zepp (only!) + paper towel (only!) for glass, soap sprayers and bug sponges for painted surfaces (wings, struts, cowl, horizontal stabilizer). (We always clean the bugs off the a/c.)

CAPF 108 is to be sent to KSWG (Attn: June), 3024 Arnold Av, Salina KS 67401. Enclose annotated fuel receipts. 108 is due five days after completion of event. (Note: KSWG does not bill for flights.) If flight is AF funded, leave a/c dry rate blank. If member funded, the rate is \$41/hr. Note on the 108 any portion paid by member.

Report squawks to ...

v1.1, 7/24/09

94°36′W

748

CAUTION: BE ALERT TO RUNWAY CROSSING CLEARANCES.

94°35′W

NC-3, 30 JUL 2009 to 27 AUG 2009

NC-3, 30 JUL 2009 to 27 AUG 2009

**BEECHCRAFT** 

**FBO** 

READBACK OF ALL RUNWAY HOLDING INSTRUCTIONS IS REQUIRED.

NC-2, 30 JUL 2009 to 27 AUG 2009